

**FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL**

**AVIATION COMPONENT SERVICES, INC.**  
**1320 EAST FRANKLIN AVENUE**  
**EL SEGUNDO, CA 90245**

**PMA NO.: PQ05347NM**  
**SUPPLEMENT NO.: 1**  
**DATE: 6/14/17**

<u>ARTICLE NAME</u>	<u>ARTICLE NUMBER</u>	<u>APPROVED REPLACEMENT FOR ARTICLE NUMBER</u>	<u>APPROVAL BASIS AND APPROVED DESIGN DATA</u>	<u>MAKE/TCH ELIGIBILITY</u>	<u>MODEL/SERIES ELIGIBILITY</u>
Attenuator Assembly – Acoustic	AC69024/ACS-13	Rolls-Royce ED101208 and/or AC69608 and/or AC69024	Test and Computations per 14 CFR § 21.303 Dwg. AC69024/ACS Rev: IR Date: 4/3/95, or later FAA approved revisions	Rolls-Royce	Rolls-Royce (RB211-535E4-37, RB211-535E4-B-37, RB211-535E4-B-75, RB211-535E4-C-37)
Air Tube	9318M28P14A CS	General Electric 9318M28P14	Test and Computations per 14 CFR § 21.303 Dwg. 9318M28P14ACS Rev: IR Date: 10/1/98, or later FAA approved revisions	General Electric Company	CF6-80C2-A1, -A2, -A3, -A5, -A8, -B1, -B2, -B4, -B6, -B1F, -B2F, -B4F, -B6F, -B6FA, -B7F, -B8F, -D1F
Air Tube	9318M28P16A CS	General Electric 9318M28P16	Test and Computations per 14 CFR § 21.303 Dwg. 9318M28P16ACS Rev: IR Date: 10/1/98, or later FAA approved revisions	General Electric Company	CF6-80C2-A1, -A2, -A3, -A5, -A8, -B1, -B2, -B4, -B6, -B1F, -B2F, -B4F, -B6F, -B6FA, -B7F, -B8F, -D1F
M Seal	1340M69P01A CS	General Electric 1340M69P01 and/or 1340M69P03	Test and Computations per 14 CFR § 21.303 Dwg. 1340M69P01ACS Rev: A Date: 5/18/02, or later FAA approved revisions	General Electric Company	CF6-80C2-A1, -A2, -A3, -A5, -A8, -A5F, -B1, -B2, -B4, -B6, -B1F, -B2F, -B4F, -B6F, -B6FA, -B7F, -D1F
W Seal	9356M78P02A CS	General Electric P/N: 9356M78P02	Test and Computations per 14 CFR § 21.303 Dwg. 9356M78P02ACS Rev: B Date: 1/31/01, or later FAA approved revisions	General Electric Company	CF6-80C2-A1, -A2, -A3, -A5, -A8, -A5F, -B1, -B2, -B4, -B6, -B1F, -B2F, -B4F, -B6F, -B6FA, -B7F, -D1F
Swirler	1292M52P07A CS	General Electric 1292M52P07	Test and Computations per 14 CFR § 21.303 Dwg. 1292M52P07ACS Rev: IR Date: 2/24/00, or later FAA approved revisions	General Electric Company	CF6-80C2-A1, -A2, -A3, -A5, -A8, -A5F, -B1, -B2, -B4, -B6, -D1F

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Y Seal	1292M81P01A CS	General Electric 1292M81P01	Test and Computations per 14 CFR §21.303 Dwg. 1292M81P01ACS Rev: IR Date: 2/2/00, or later FAA approved revisions	General Electric Company	CF6-80C2-A1, -A2, - A3, -A5, -A8, -A5F, -B1, -B2, -B4, -B6, -B1F, -B2F, -B4F, -B6F, -B6FA, -B7F, -D1F
W Seal	9356M78P01A CS	General Electric 9356M78P01	Test and Computations per 14 CFR § 21.303 Dwg. 9356M78P01ACS Rev: IR Date: 6/19/01, or later FAA approved revisions	General Electric Company	CF6-80A, -A2
Seal Assembly - Fire	314W5510- 67ACS	Boeing 314W5510-67	Test and Computations per 14 CFR § 21.303 Dwg. 314W5510- 67ACS Rev: A Date: 2/6/03, or later FAA approved revisions	Boeing	777-200, -300
Seal Assembly - Fire	314W5510- 69ACS	Boeing 314W5510-69	Test and Computations per 14 CFR § 21.303 Dwg. 314W5510- 67ACS Rev: A Date: 2/6/03, or later FAA approved revisions	Boeing	777-200, -300

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Seal-External Pressure	1359M84P08A CS	1359M84P06, 1359M84P08, 1359M84P11	Test and Computations per 14 CFR § 21.303 Dwg No: 1359M84P08ACS Rev: IR Date:10/25/2004, or later FAA approved revisions	CFM International, S.A.	CFM56-5C3/P, CFM56-5C4, CFM56-5C4/P, CFM56-5C4/1, CFM56-5C4/1P, CFM56-7B20, CFM56-7B22, CFM56- 7B22/B1, CFM56-7B24, CFM56-7B24/B1, CFM56- 7B26, CFM56-7B27, CFM56- 7B26/B1, CFM56-7B27/B1, CFM56-7B27/B3, CFM56- 7B30/2, CFM56-75B22/2, CFM56-7B24/2, CFM56- 7B26/2, CFM56-7B27/2, CFM56-7B27A, CFM56- 7B26/B2, CFM56-7B20/3, CFM56-7B22/3, CFM56- 7B22/3B1, CFM56-7B24/3, CFM56-7B24/3B1, CFM56- 7B26/3, CFM56-7B26/3B1, CFM56-7B26/3B2, CFM56- 7B27/3, CFM56-7B27/B1, CFM56-7B27/3B3, CFM56- 7B26/3F, CFM56-7B26/3B2F, CFM56-7B27/3F, CFM56- 7B27/3B1F. CFM56-7B27A/3



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Seal-External Pressure	1359M84P08A CS	1359M84P06, 1359M84P08, 1359M84P11	Test and Computations per 14 CFR § 21.303 Dwg No: 1359M84P08ACS Rev: IR Date:10/25/2004, or later FAA approved revisions	CFM International, S.A.	CFM56-5-A1/F, CFM56-5A3, CFM56-5A4, CFM56-5A4/F, CFM56-5A5, CFM56-5A5/F, CFM56-5B1, CFM56-5B1/P, CFM56-5B1/2P, CFM56-5B2, CFM56-5B2/P, CFM56- 5B2/2P, CFM56-5B3/P, CFM56-5B3/P1, CFM56- 5B3/2P1, CFM56-5B4, CFM56-5B4/P, CFM56- 5B4/P1, CFM56-5B4/2P, CFM56-5B4/2P1, CFM56-5B5, CFM56-5B5/P, CFM56-5B6, CFM56-5B6/P, CFM56- 5B6/2P, CFM56-5B7, CFM56- 5B7/P, CFM56-5B8/P, CFM56-5B9/P, CFM56-5C2, CFM56-5C2/4, CFM56-5C2/F, CFM56-5C2/F4, CFM56- 5C2/G, CFM56-5C2/G4, CFM56-5C2/P, CFM56-5C3/F, CFM56-5C3/F4, CFM56- 5C3/G, CFM56-5C3/G4
Box, Trim EGT	BC107-01ACS	CFM International, 301-813-902-0; Auxitrol,BC107 -01	Test and Computation sper 14 CFR § 21.303 Dwg No: BC107- 01ACS Rev: IR Date: 3/28/05, or later FAA approved revisions	CFM International	CFM56-3B1, CFM56-3B2, CFM56-3C1
CFM56- 7BEGT Harness	TC296-03ACS	CFM International S.A. TC296-02, TC296-06, TC296-04	Test and Computation s per 14 CFR § 21.303 Master Drawing List No: TC296-03ACS Rev: IR Date: 3/28/05, or later FAA approved revisions	CFM International S.A.	CFM56-7B18, CFM56-7B20, CFM56-7B20/2, CFM56-7B22, CFM56-7B22/2, CFM56- 7B22/B1, CFM56-7B22/B2, CFM56-7B24, CFM56-7B24/2, CFM56-7B24/B1, CFM56- 7B26, CFM56-7B26/2, CFM56-7B26/B1, CFM56- 7B26/B2, CFM56-7B27, CFM56-7B27/2, CFM56- 7B27/B1, CFM56-7B27/B3

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Wiring Harness T495- 2TC	TC272-01ACS	CFM International, S.A. TC272-00, TC272-01, TC272-02	Test and Computations per 14 CFR § 21.303 Master Drawing List No: TC272-01ACS Rev: IR Date: 9/29/05, or later FAA approved revisions	CFM International, S.A.	CFM56-5C2, CFM56-5C2/4, CFM56-5C2/F, CFM56- 5C2/F4, CFM56-5C2/G, CFM56-5C2/G4, CFM56- 5C2/P, CFM56-5C CFM56- 5C3/G, CFM56-5C3/G4, CFM56-5C3/P, CFM56-5C4, CFM56-5C4/P, CFM56-5C4/1, CFM56-5C4/1P
Wiring Harness T495- 2TC	TC273-01ACS	CFM International, S.A TC273-00, TC273-01, TC273-02	Test and Computations per 14 CFR § 21.303 Master Drawing List No: TC273-01ACS Rev: IR Date: 9/29/05, or later FAA approved revisions	CFM International, S.A.	CFM56-5C2, CFM56-5C2/4, CFM56-5C2/F, CFM56- 5C2/F4, CFM56-5C2/G, CFM56-5C2/G4, CFM56- 5C2/P, CFM56-5C CFM56- 5C3/G, CFM56-5C3/G4, CFM56-5C3/P, CFM56-5C4, CFM56-5C4/P, CFM56-5C4/1, CFM56-5C4/1P
Lead Assembly, Electrical L.H.	TC174-01ACS	CFM International TC174-00, TC174-01, TC174-02	Test and Computations per 14 CFR § 21.303 ACS MDL TC174- 01ACS Rev: IR Date: 09/28/05, or later FAA approved revisions	CFM International, S.A.	CFM56-2A, CFM56-2A-2, CFM56-2A-3, CFM56-2B, CFM56-2B-1, CFM56-3, CFM56-3-B1, CFM56-3B, CFM56-3B-2, CFM56-3C, CFM56-3C-1
Lead Assembly, Electrical L.H.	TC200-01ACS	CFM International TC200-00, TC200-01, TC200-02	Test and Computations per 14 CFR § 21.303 ACS MDL TC200- 01ACS Rev: IR Date: 09/28/05, or later FAA approved revisions	CFM International, S.A.	CFM56-2A, CFM56-2A-2, CFM56-2A-3, CFM56-2B, CFM56-2B-1, CFM56-3, CFM56-3-B1, CFM56-3B, CFM56-3B-2, CFM56-3C, CFM56-3C-1



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Wiring Harness, T495-2TC	TC315-01ACS	CFM International, S.A. TC315-00, TC315-01, TC315-02	Test and Computation per 14 CFR § 21.303 Dwg. ACS MDL TC315-01ACS Rev: IR Date:11/13/06, or later FAA approved revision	CFM International, S.A.	CFM56-5-A1, CFM56-5A3, CFM56-5A4, CFM56-5A5, CFM56-5-A1/F, CFM56- 5A4/F, CFM56-5A5/F, CFM56-5B1, CFM56-5B1/P, CFM56-5B1/2P, CFM56- 5B3/P, CFM56-5B3/P, CFM56-5B3/2P, CFM56- 5B3/2P, CFM56-5B4, CFM56- 5B4/P, CFM56-5B4/P1, CFM56-5B4/2P, CFM56- 5B4/2P1, CFM56-5B5, CFM56-5B5/P, CFM56-5B6, CFM56-5B6/P, CFM56-5B7/P, CFM56-5B8/P, CFM56-5B9/P, CFM56-5B9/2P

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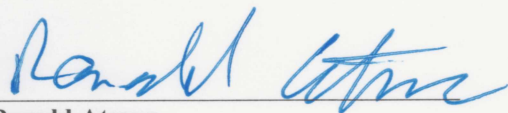
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Wiring Harness, T495-3TC	TC316-01ACS	CFM International, S.A. TC316-00, TC316-01, TC316-02	Test and Computations per 14 CFR § 21.303 Dwg. ACS MDL TC316-01ACS Rev: IR Date: 11/13/06, or later FAA approved revisions	CFM International, S.A.	CFM56-5-A1, CFM56-5A3, CFM56-5A4, CFM56-5A5, CFM56-5-A1/F, CFM56-5A4/F, CFM56-5A5/F, CFM56-5B1, CFM56-5B1/P, CFM56-5B1/2P, CFM56-5B3/P, CFM56-5B3/P, CFM56-5B3/2P, CFM56-5B3/2P, CFM56-5B4, CFM56-5B4/P, CFM56-5B4/P1, CFM56-5B4/2P, CFM56-5B4/2P1, CFM56-5B5, CFM56-5B5/P, CFM56-5B6, CFM56-5B6/P, CFM56-5B7/P, CFM56-5B8/P, CFM56-5B9/P, CFM56-5B9/2P

-----End of Data-----

## GENERAL NOTES:

Provide minor design changes in a manner as determined by the ACO. Provide major design changes to drawings and specifications in the same manner as that for an original FAA-PMA.

The FAA accepted the ICA approach for the above articles with their designs. These ICA may refer to those of the respective articles from the holders of type certificates. Otherwise, provide supplemental ICA for differences in the replacement articles. Make referral statements or supplemental ICA readily available per 14 CFR 21.50.



Ronald Atmur  
Manager, Propulsion Branch  
Los Angeles Aircraft Certification Office



Tara Frame  
Aviation Safety Inspector  
Los Angeles Manufacturing Inspection District Office